



# Cost Reduction for Advanced Integration Heat Exchanger Technology for Microreactors

NEUP Project 21-24226



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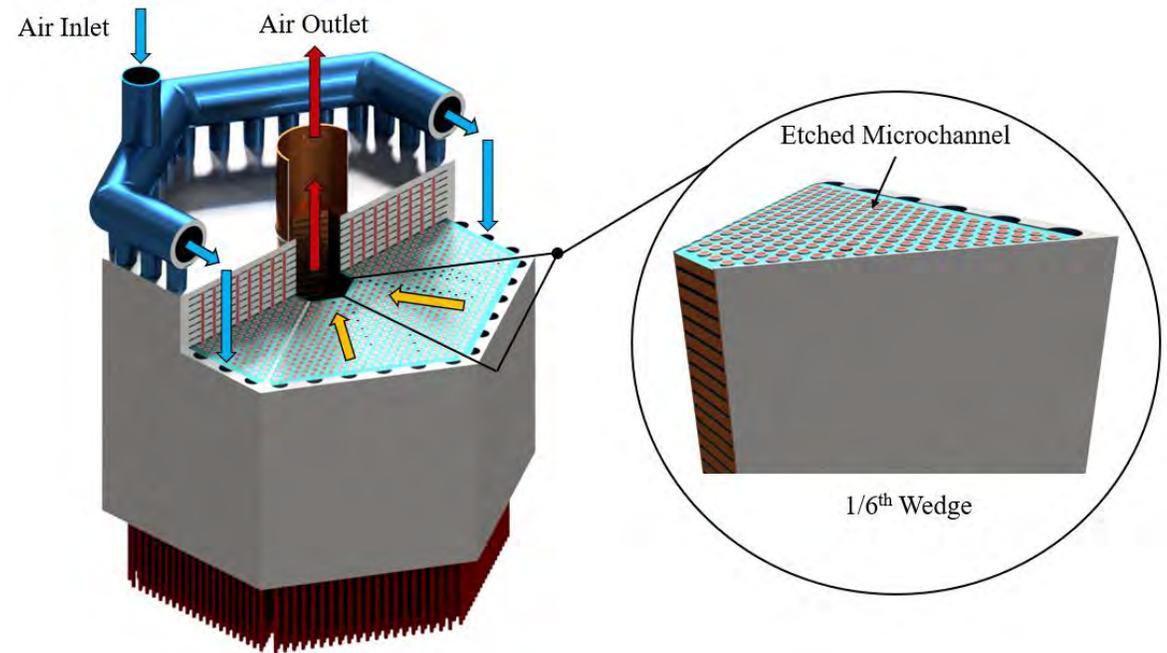


February 27, 2025

# Presentation Overview



- Project background and recap
- Air Brayton testing
- sCO<sub>2</sub> optimization
- Conclusions and future work





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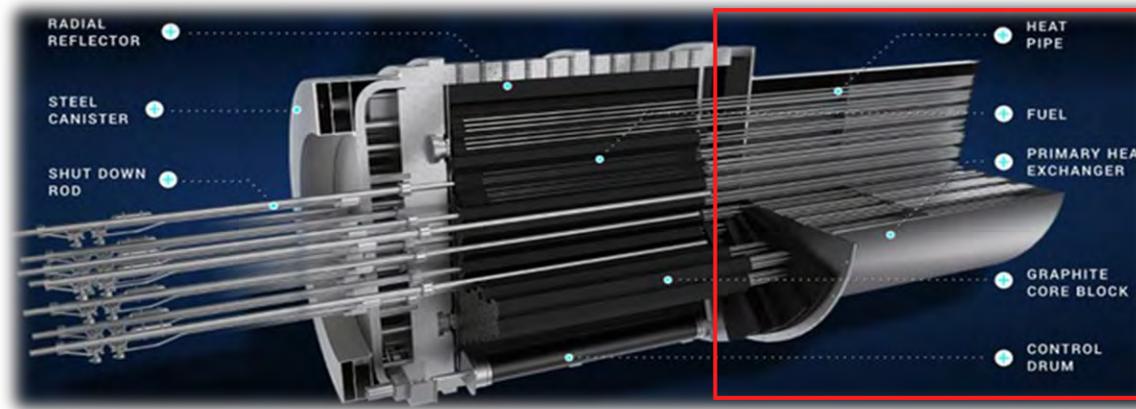
# Project Background

# Interface Heat Exchanger



- Objectives

- Development and validation of microreactor integration heat exchanger design tools
- Demonstrate potential cost-reduction/performance improvements in the context of an eVinci™-like microreactor
- Obtain benchmark and validation data
- Demonstrate sub-size PCHE-based integration HX for sCO<sub>2</sub> and air working fluids
- Train several students for nuclear industry



eVinci™ Micro-Reactor, Courtesy of Westinghouse Electric Company LLC

# Printed Circuit Heat Exchanger

- Printed circuit heat exchanger
  - Thin metal sheets are chemically etched
  - Diffusion bonded together
  - Forms microchannels with high heat transfer area
- HPIHX PCHE
  - Single fluid, cross-flow
  - Add heat pipe holes to plates for HPIHX
  - As interlayers plates become very thin PCHE approaches cross-flow heat exchanger



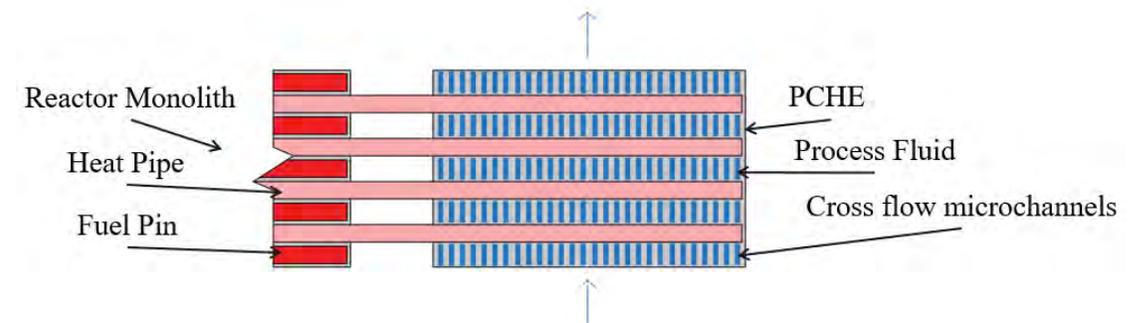
Chemically etched microchannel [1]



Cutout of diffusion bonded PCHE (VPEI [2])



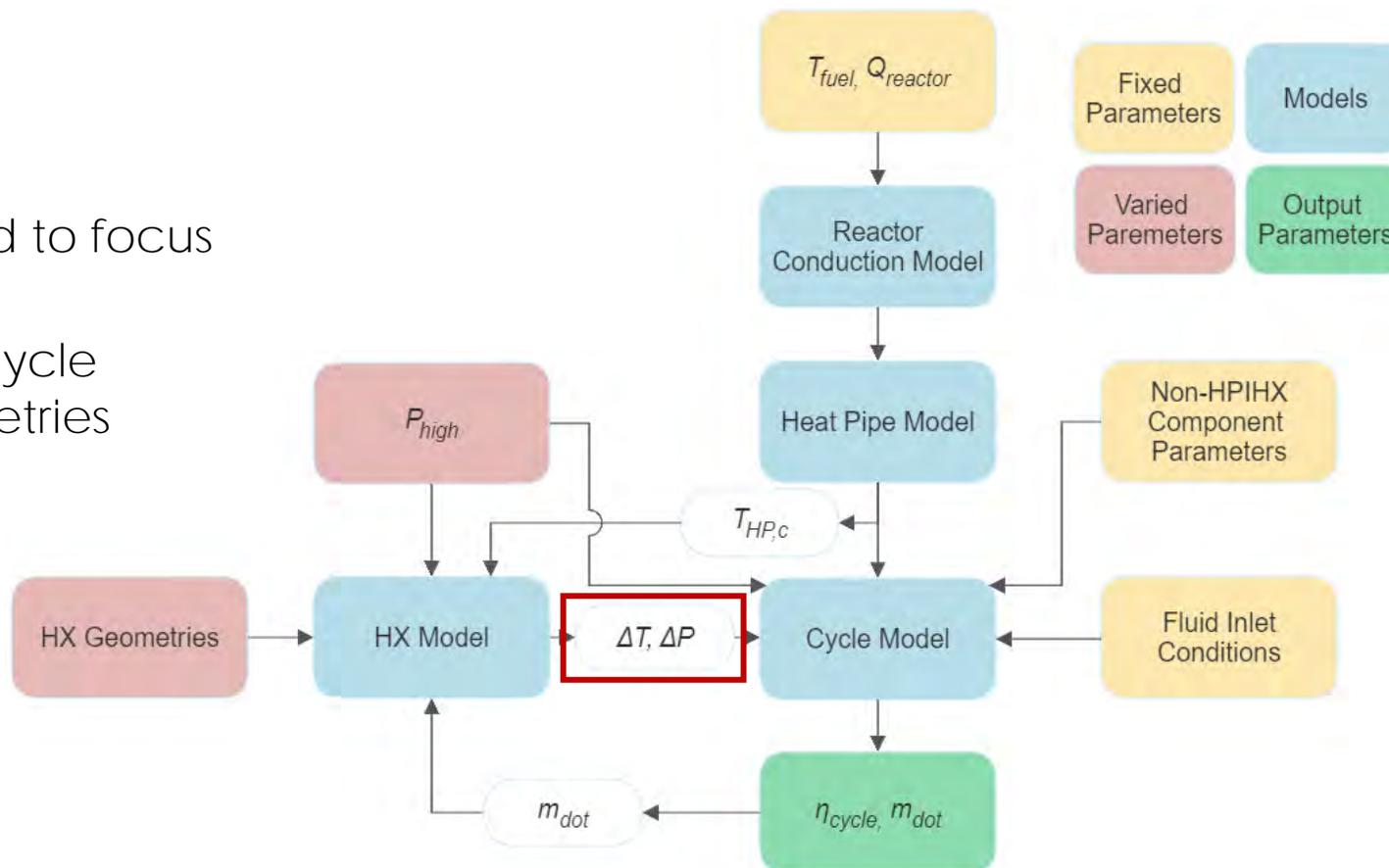
PCHE-based HPIHX





# Interface Heat Exchanger Optimization

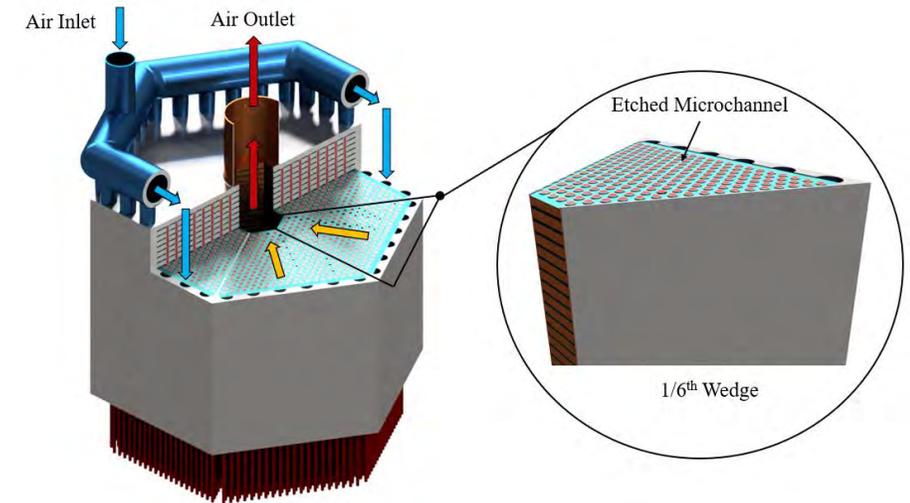
- Optimized using cycle model
  - Reactor conduction model
  - Heat pipe model
  - Heat exchanger models
  - All non-HPIHX parameters are fixed to focus analysis on the heat exchanger
  - Vary HX geometries to maximize cycle efficiency and find optimal geometries
- AFHX and PCHE HX models
- $\Delta P = P_{in} - P_{out}$
- $\Delta T = T_{HX_{max}} - T_{fluid,out}$



# Project Organization



- Air Brayton modeling
  - Develop reactor, HX, and cycle models
  - Optimize air Brayton HX
- Performance demonstration
  - ➔ ◦ Design and manufacture air test specimen
  - Demonstrate performance with N<sub>2</sub> at MAGNET
  - ➔ ◦ Demonstrate performance with N<sub>2</sub> at UW
- sCO<sub>2</sub> modeling and testing
  - Optimize sCO<sub>2</sub> Brayton HX
  - ➔ ◦ Design and manufacture sCO<sub>2</sub> test specimen
  - Demonstrate performance with sCO<sub>2</sub> at UW





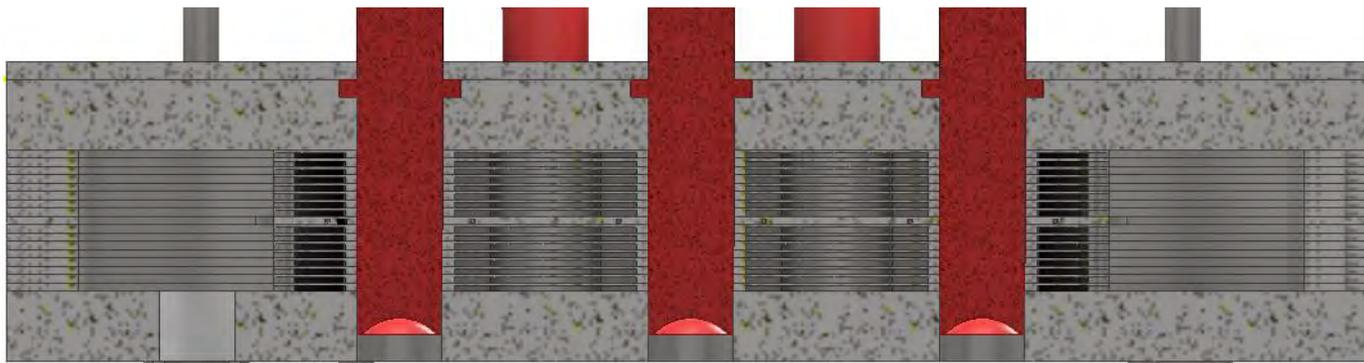
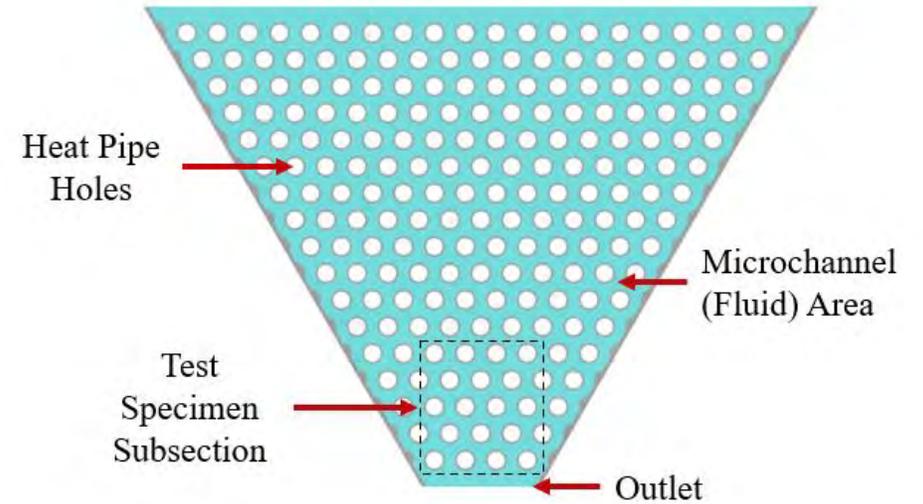
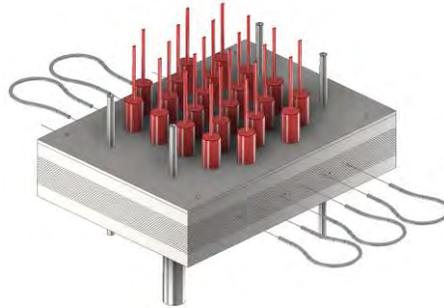
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# Air Brayton Testing

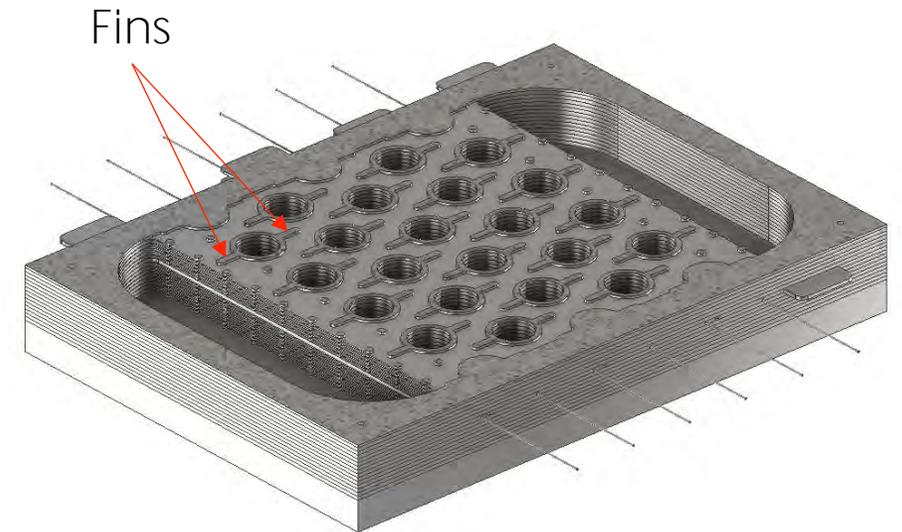
Facility, testing results, validated model

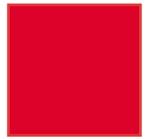
# Air Brayton Test Specimen

- Diffusion bonded
  - 16 -1.5 mm 316 SS plates
  - 1- instrumentation layer
  - 10" x 7" x 5"
- 22-125 W cartridge heaters
- Subsection replicates conditions of the full "wedge"



	Test Article	Full Size HX
Power	2.75 kW	5000 kW
Energy density	63 W/in <sup>2</sup>	67 W/in <sup>2</sup>
Cross section	0.08 in <sup>2</sup>	0.08-0.34 in <sup>2</sup>
Mass flow rate	0.02-0.08 kg/s	16.4 kg/s





# Air Brayton Test Specimen



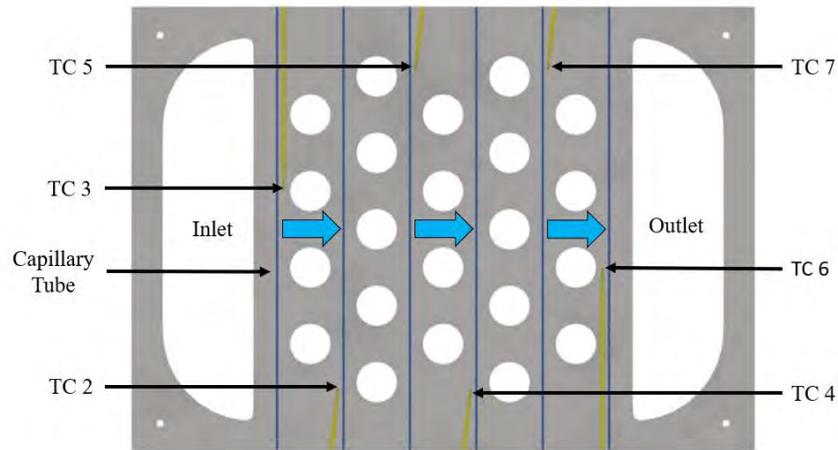
- Instrumentation
  - 4-pressure taps (2- $\Delta P$  measurements)
  - 6-TC probes in instrument layer
  - 6-embedded capillary tubes for fiber optics
  - $T_{out}$  initially located in the header



Cartridge heaters

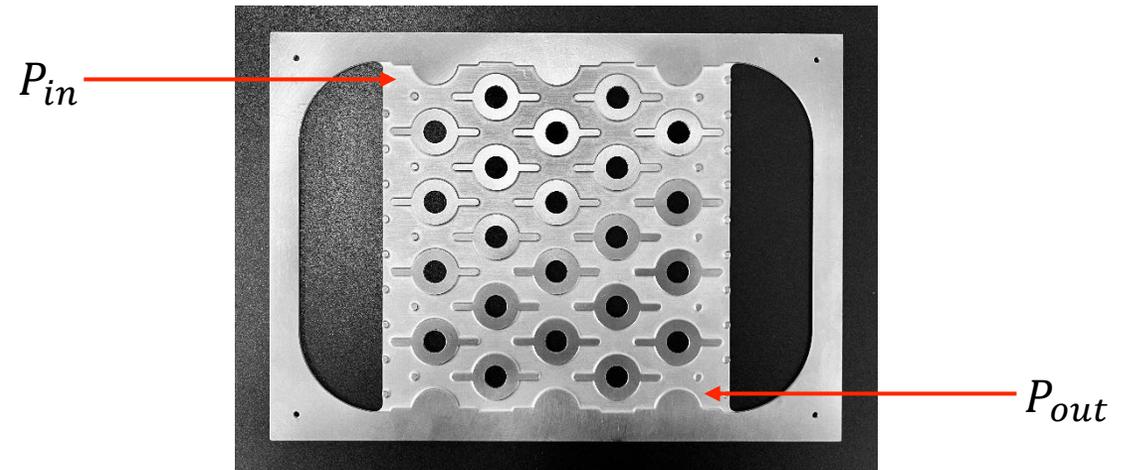


Air test specimen



$$\Delta T = T_{hp,max} - T_{fluid,out} = TC_6 - T_{fluid,out}$$

$$\Delta P_{channel} = P_{in} - P_{out}$$

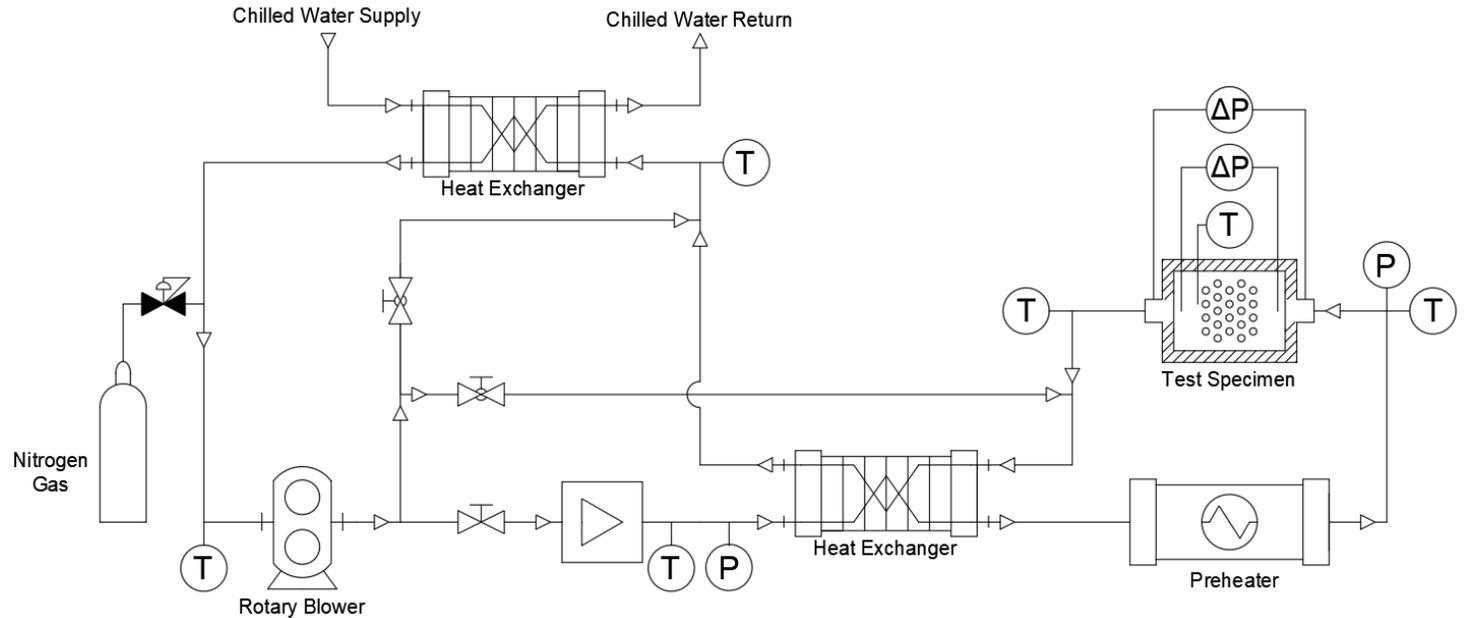


Air test flow channel

# UW Nitrogen Loop



- UW designed operating conditions
  - $\dot{m} \sim 0.02\text{-}0.15$  kg/s
  - $T_{in} \sim 40\text{-}450^\circ\text{C}$
  - $P_{in} \sim 150\text{-}1400$  kPa
- Test Matrix
  - $\dot{m} = 0.02\text{-}0.1$  kg/s
  - $T_{in} = 100\text{-}450^\circ\text{C}$
  - $P_{in} = 400, 550, 700$  kPa
  - $\dot{q} = 690, 1380, 2060, 2780$  W



UW-MAGNET HPIHX Schematic

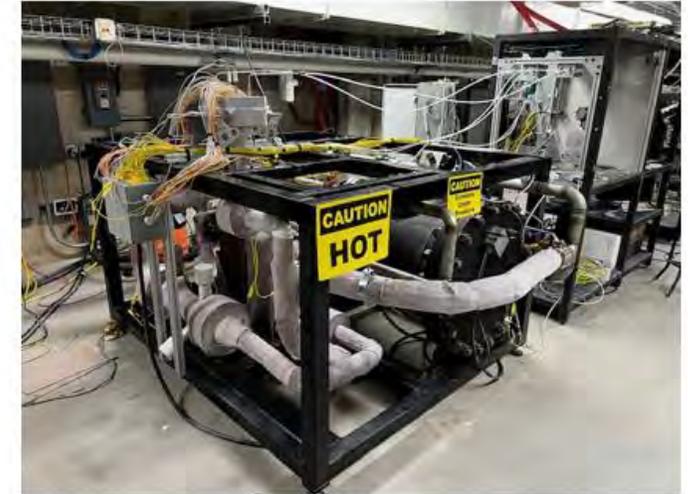
# UW Nitrogen Loop



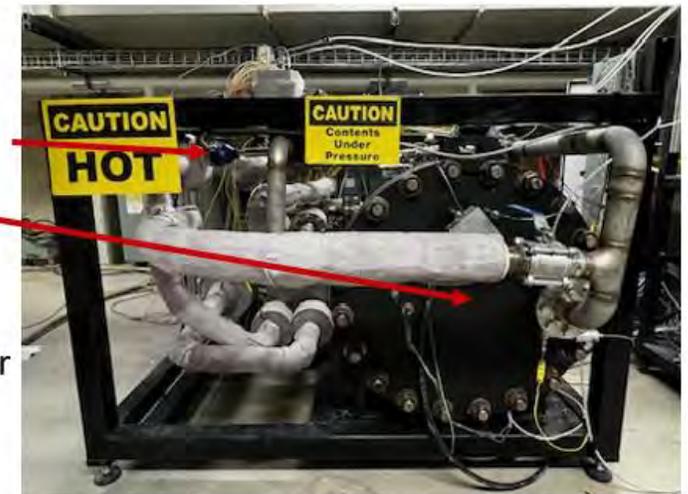
- Blower
  - Gardner Denver,
- Motor
  - BlackMax, 11 kW
- Flow meter
  - Vortex Shedder
- Recuperator & chiller
  - Kelvion, brazed plate
- Preheater
  - Osram, 20 kW
- Controls & DAQ
  - LabView
  - NI compactRIO system



N2 Test specimen



Chiller



Flow meter

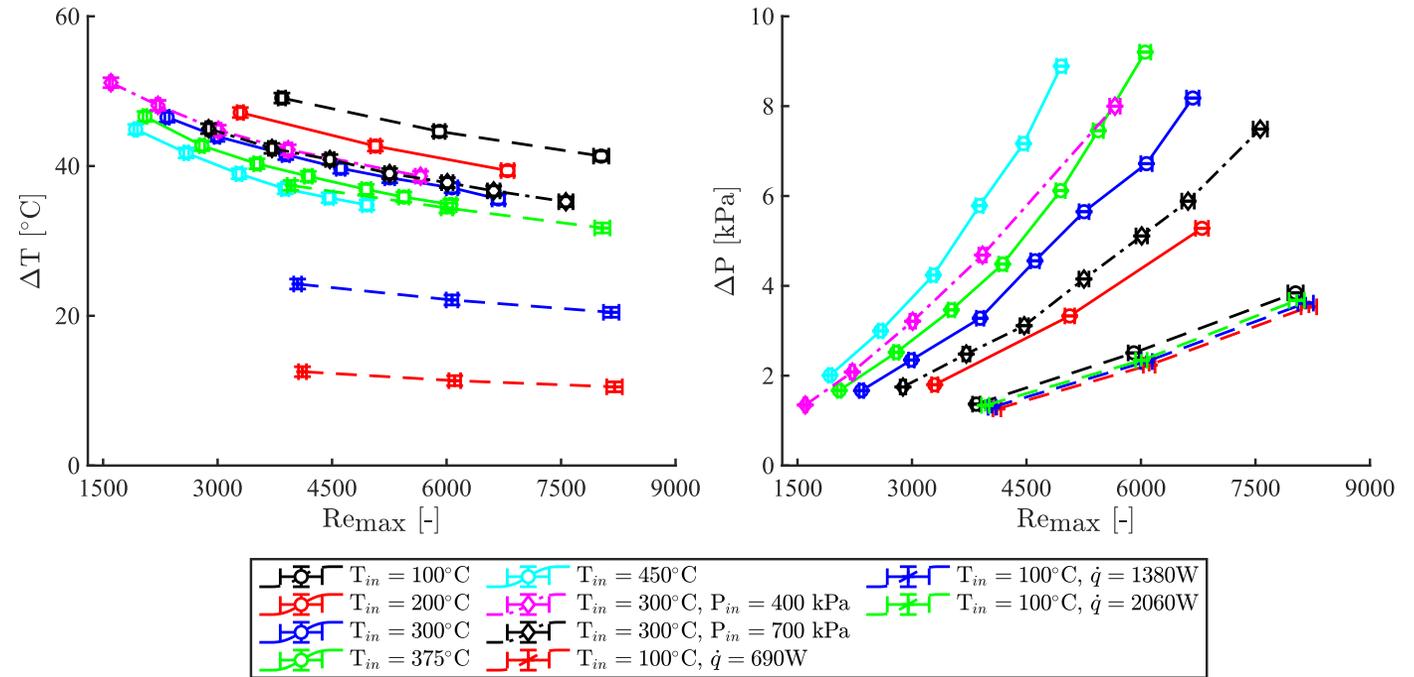
Blower

Recuperator  
Heater



# Testing Results

- Approach temperature vs. Reynolds
  - Increased  $\dot{q}$  → increased  $\Delta T$
  - Increased  $T_{in}$  → decreased  $\Delta T$
  - Varied  $P_{in}$  ~ no impact
- Pressure drop vs. Reynolds
  - Varied  $\dot{q}$  ~ no impact
  - Increased  $T_{in}$  → increased  $\Delta P$
  - Increased  $P_{in}$  → decreased  $\Delta P$



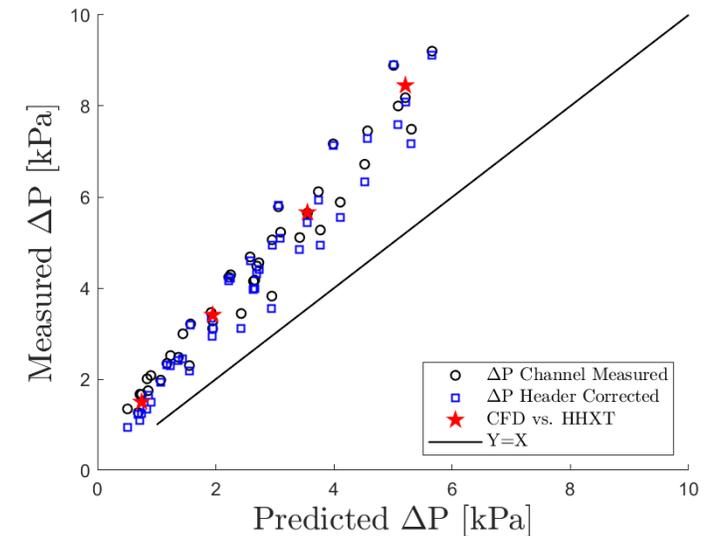
Nitrogen test results

# Experiment vs. Model

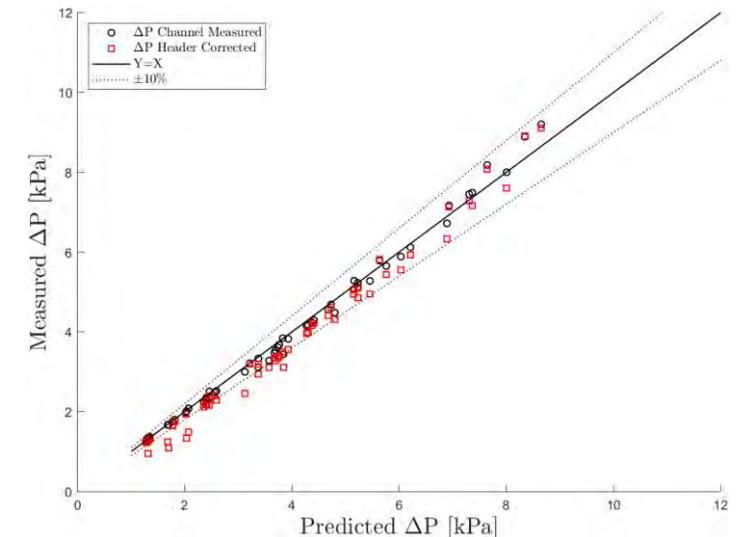


- Pressure drop test vs. model
  - Initial model (top) underpredicted  $\Delta P$ 
    - Colebrook friction factor
  - Calculated average friction factor
    - $\bar{f} = \frac{2\Delta P_{channel} D_h}{\rho v^2 L}$
  - Friction factor correlation was generated for HHXT model
    - $\bar{f}_{HHXT} = a Re_{max}^b$
    - $b$  was fixed to maintain behavior
    - $a$  was varied to minimize model error
  - Updated model (bottom) predicted experimental pressure drop within 10%

Initial Model



Updated Model

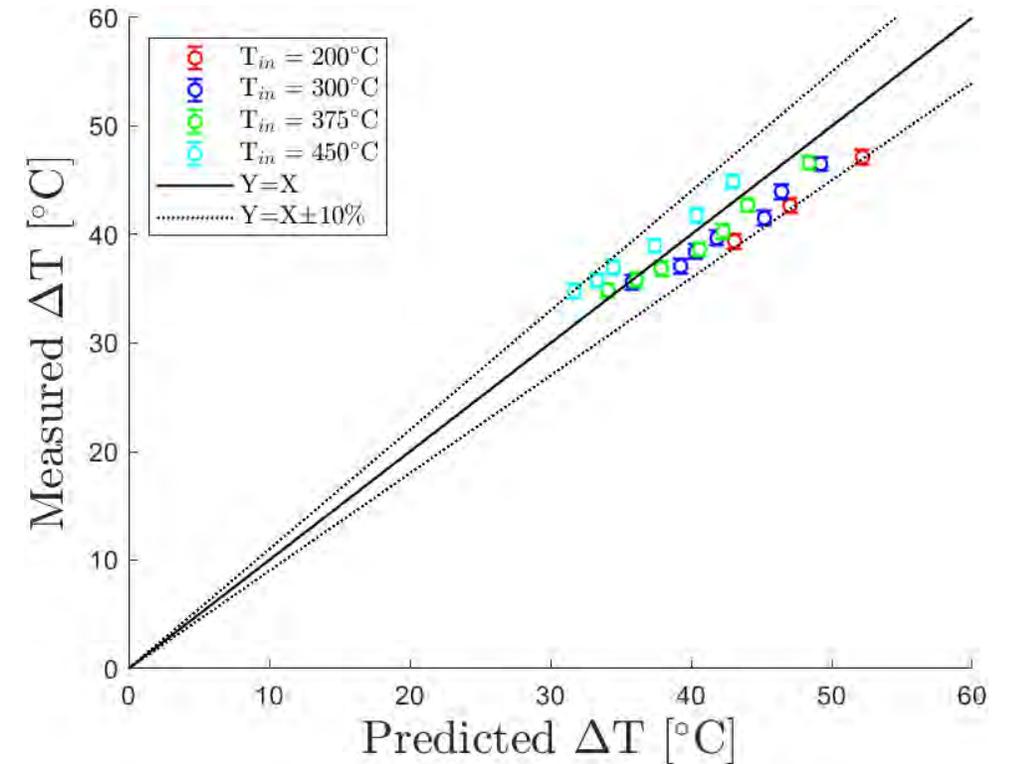


# Approach Temp. Test vs. Model



- Initial results showed general agreement within 15%
- Attributed the spread at different temperatures to heat losses
- Accounting for heat losses at elevated temperatures collapsed the points
- Relatively small change to the data results in trends agreeing between model and experiment

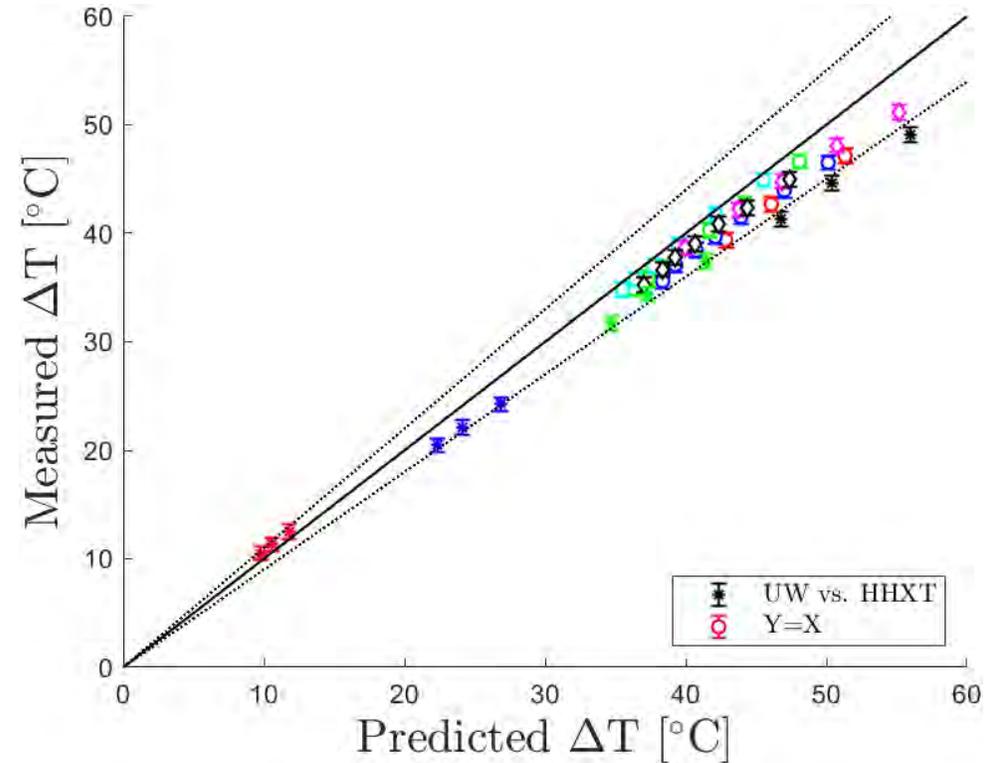
Initial experiment vs. model



# Experiment vs. Model



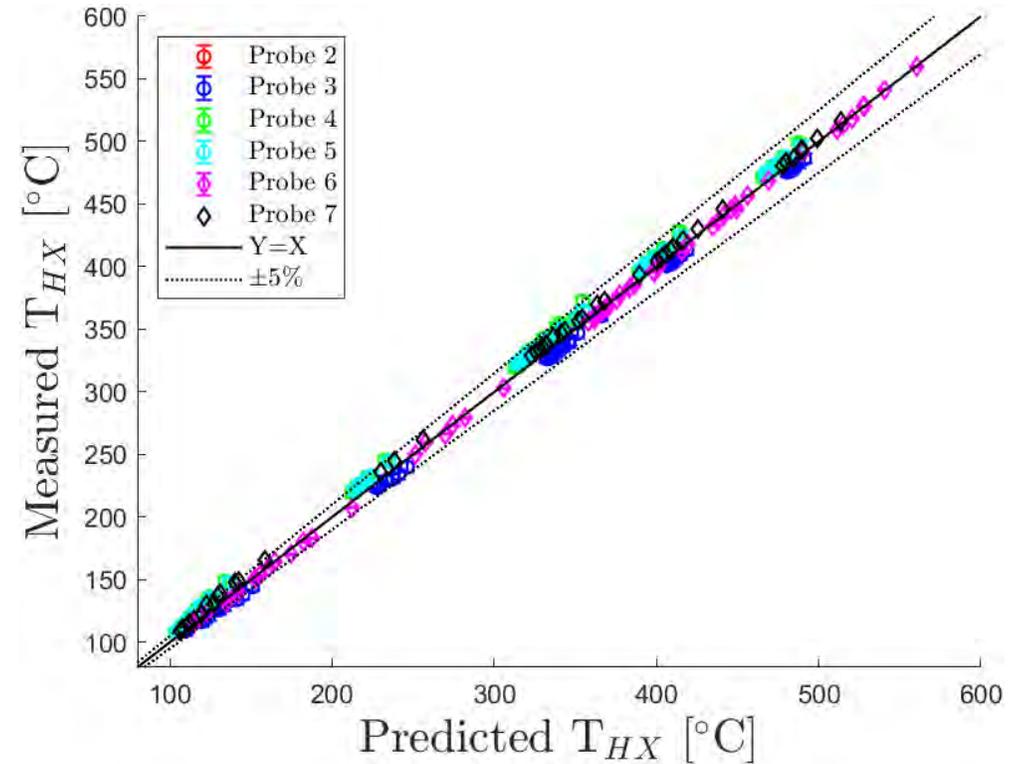
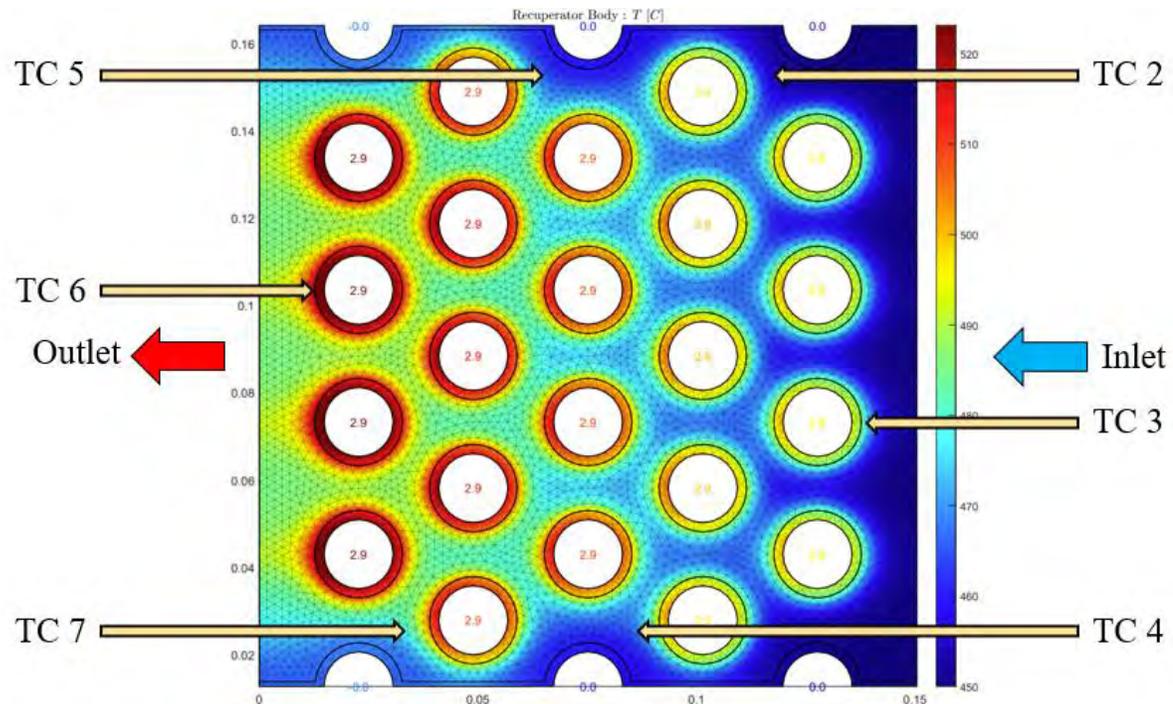
- Approach temperature test vs. model
  - After correction for heat losses all but 3 predicted values within 10% of experiment
  - Overprediction in general, partially due to fin geometry not being resolved by the model



# Experiment vs. Model



- Temperature probes versus model
  - Generally, fall within 5% of predicted value
  - TC 3 and TC 6 very close agreement (heater wall)
  - TC 2, 4, 5, 7 have fin effect from increased layer thickness of instrument layer ~ (2.5 mm versus 0.5 mm)

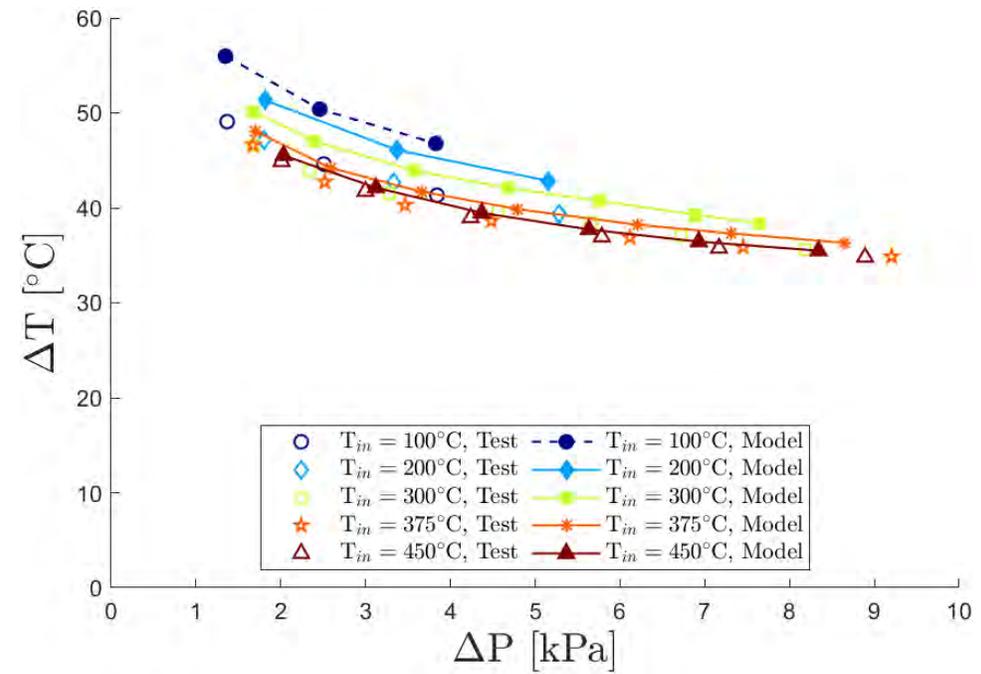
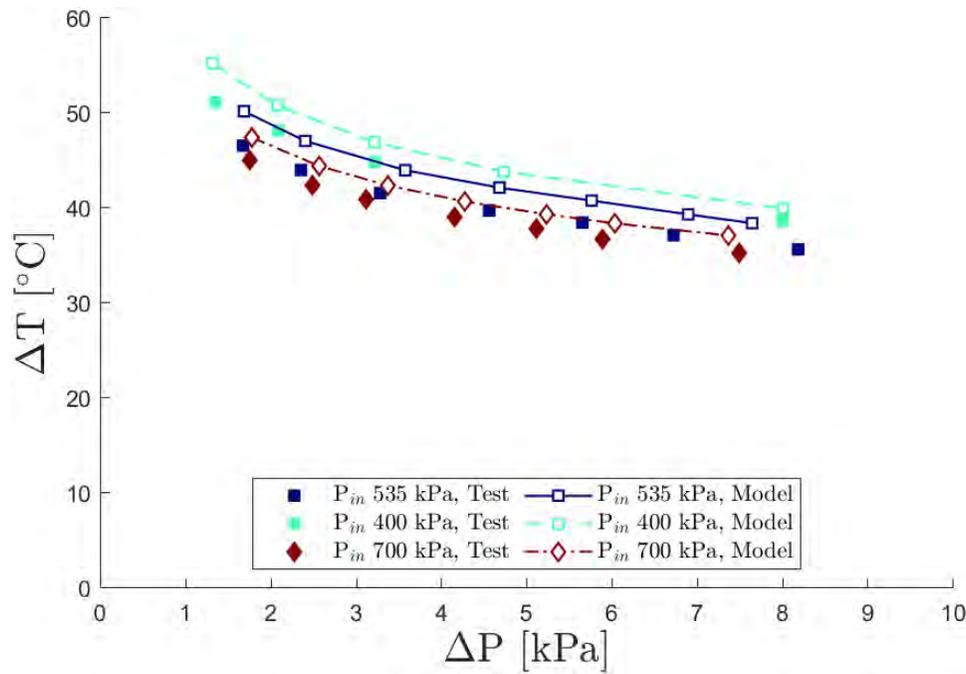


# Experiment vs. Model



- Test and model,  $\Delta T$  vs.  $\Delta P$ 
  - Varied pressures
  - Trends show agreement
- Model validation → cycle model

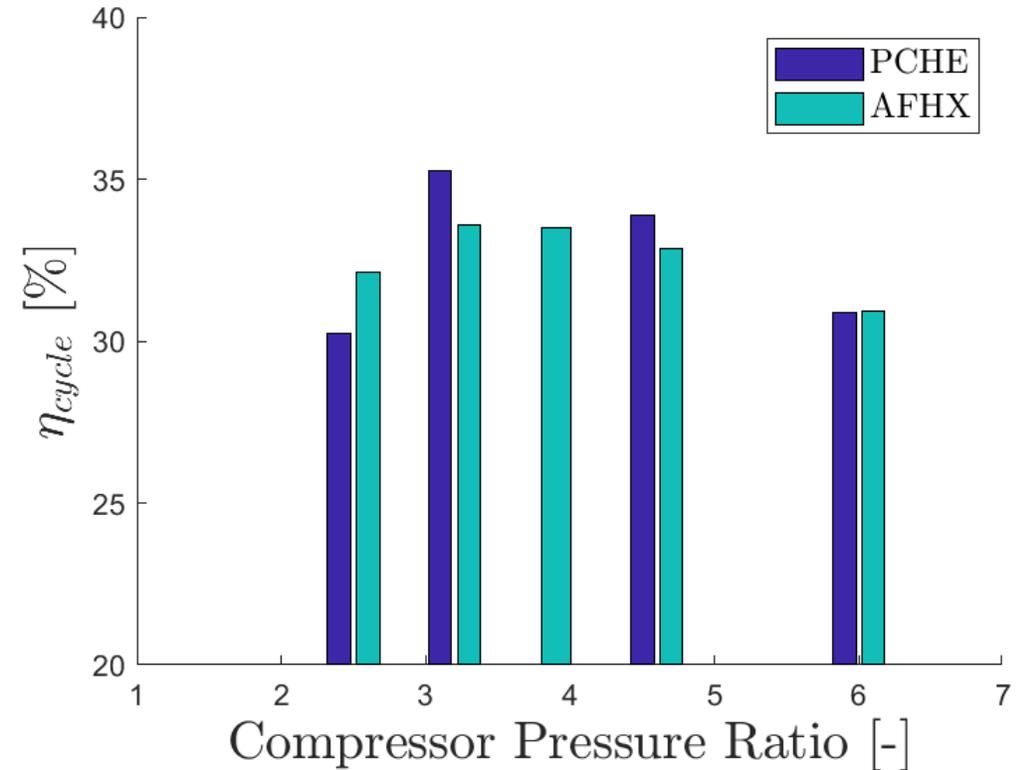
- Test and model,  $\Delta T$  vs.  $\Delta P$ 
  - Varied temperature
  - Error (model overpredicts) as  $T_{in} \downarrow$  indicates the predicted performance is conservative



# Validated Cycle Model Results



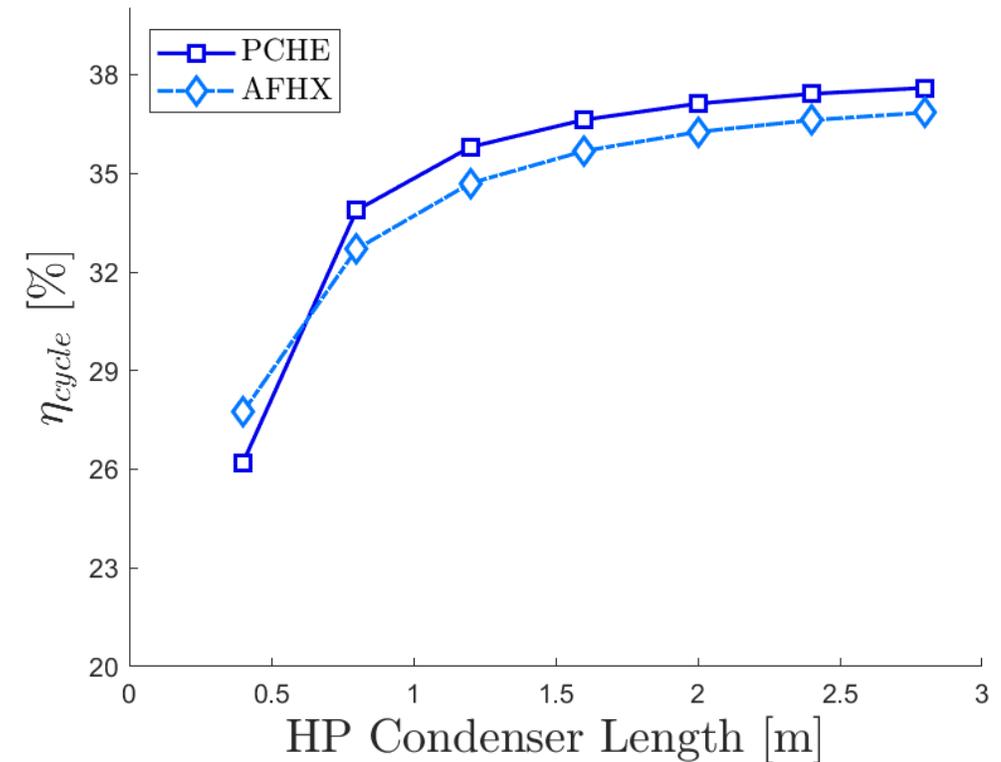
- Updated HHXT model
  - New friction factor correlation
  - General increase in pressure drop
  - Slight shift of previous optimal and decrease in efficiency (2.2% → 1%)
- AFHX and PCHE comparison
  - PCHE – 33.9% (460 kPa)
  - AFHX – 32.9% (460 kPa)
  - PCHE – 35.3% (320 kPa)
  - AFHX – 33.6% (320 kPa)





# Heat Pipe Length Study

- Varied heat pipe condenser length (460 kPa)
  - Limit near 38% as condenser length increases
  - Below 0.8 m performance and cycle efficiency has significant decrease
- 3.2% increase for PCHE (0.8 m to 2.0 m)
- 3.6% increase for AFHX (0.8 m to 2.0 m)
- At 2.8 m
  - PCHE – 37.6%
  - AFHX – 36.9%
- Westinghouse has manufactured heat pipes up to 4 m





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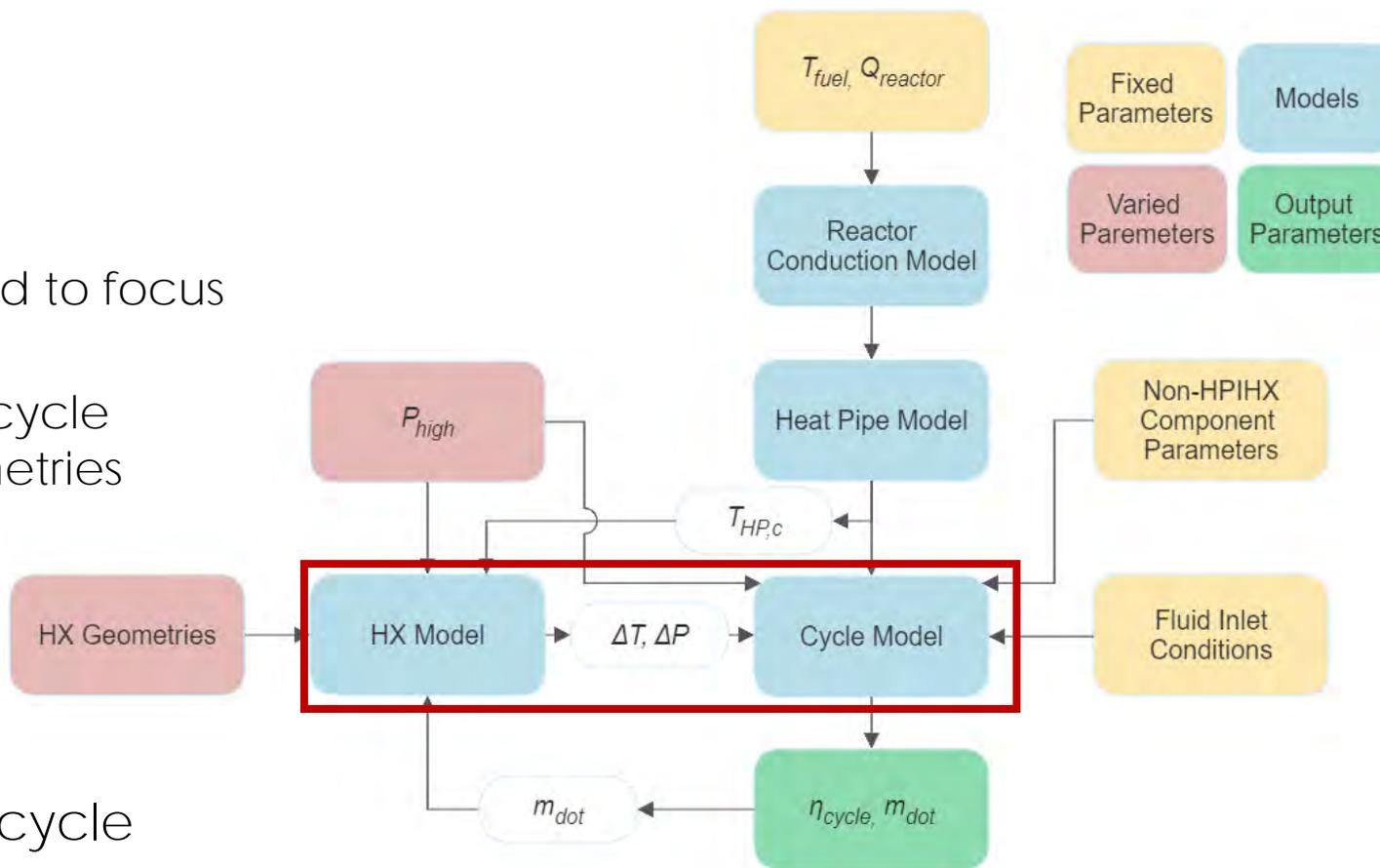
# sCO<sub>2</sub> Optimization

Model, results, test article



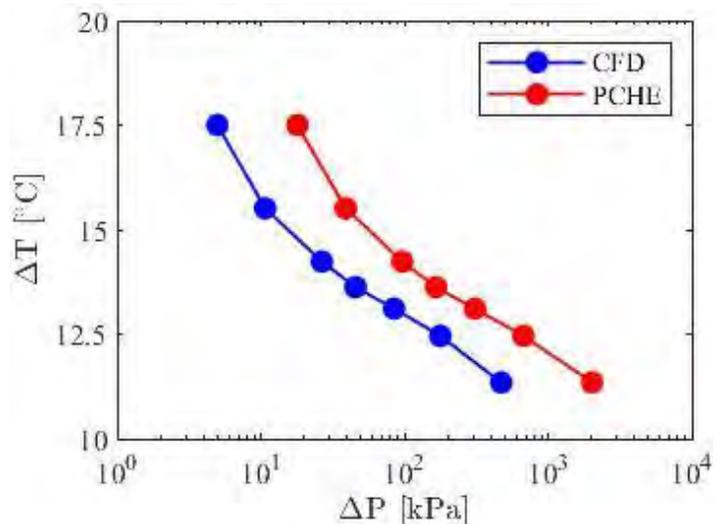
# Interface Heat Exchanger Optimization

- Optimized using cycle model
  - Reactor conduction model
  - Heat pipe model
  - Heat exchanger models
  - All non-HPIHX parameters are fixed to focus analysis on the heat exchanger
  - Vary HX geometries to maximize cycle efficiency and find optimal geometries
- AFHX and PCHE HX models
- $\Delta P = P_{in} - P_{out}$
- $\Delta T = T_{hp_{max}} - T_{fluid,out}$
- Air Brayton cycle  $\rightarrow$  sCO<sub>2</sub> Brayton cycle
- HHXT PCHE model  $\rightarrow$  CFD PCHE model

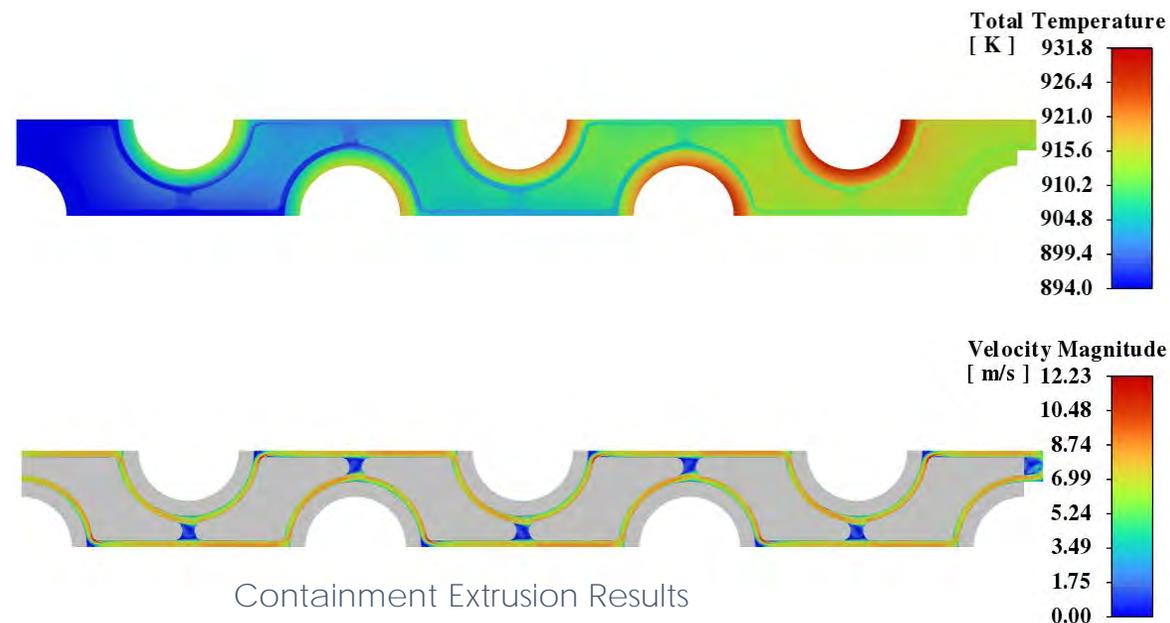


# CFD Unit Cell Results

- Further refinement of best geometry
  - Channel thickness varied:  $th_{gap} = 0.25 - 1.5$  [mm]
  - Openings to prevent full channel blockage
- Unit cell pressure drop used to calculate full sized PCHE pressure drop for cycle model



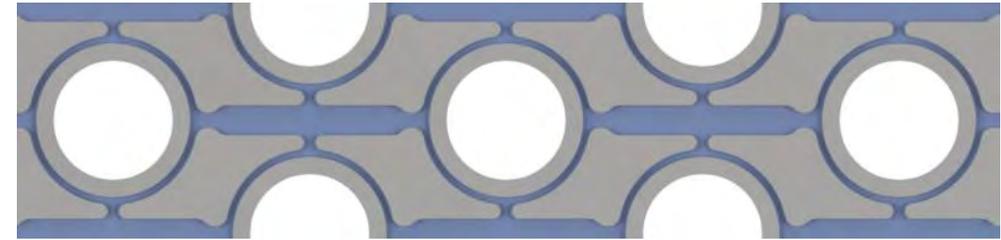
CFD unit cell approach temperature and pressure drop



# sCO<sub>2</sub> Cycle Analysis

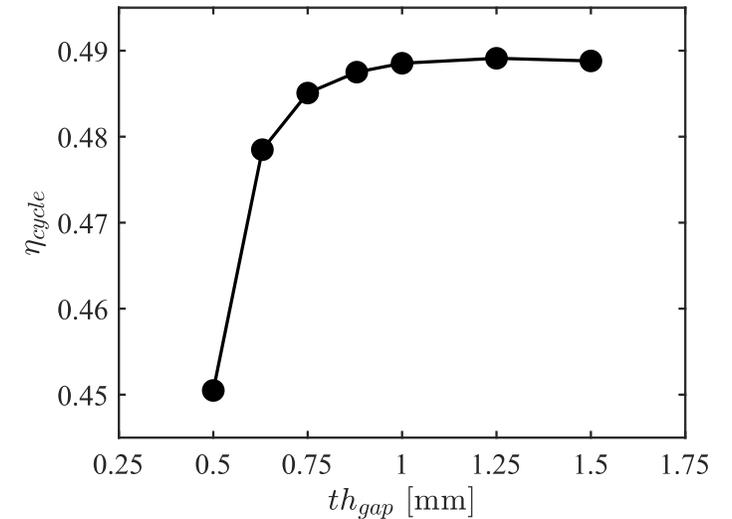


- Incorporated PCHE model into sCO<sub>2</sub> cycle
  - Solved steady-state pressure and temperature across PCHE
  - Optimized channel thickness:  **$th_{gap,opt} = 1.25 \text{ mm}$**
- sCO<sub>2</sub> Brayton cycle efficiency improved 13.3% over the air Brayton cycle



Optimized geometry

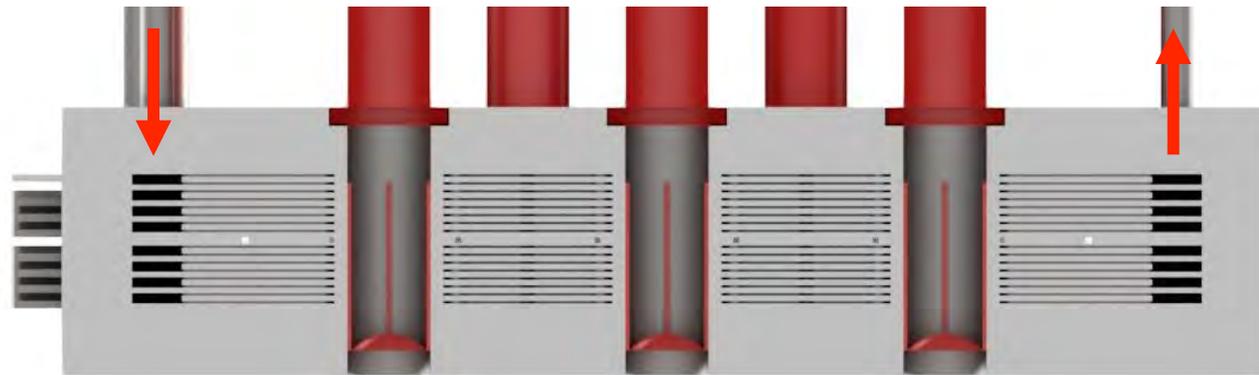
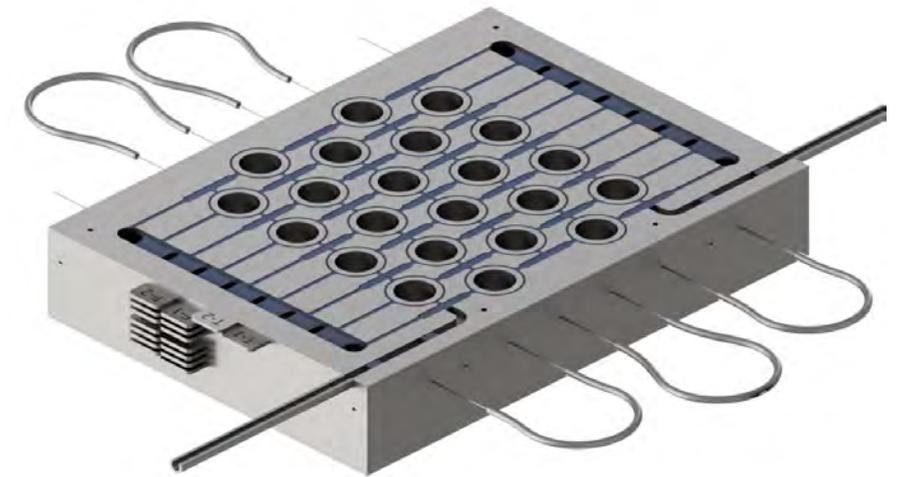
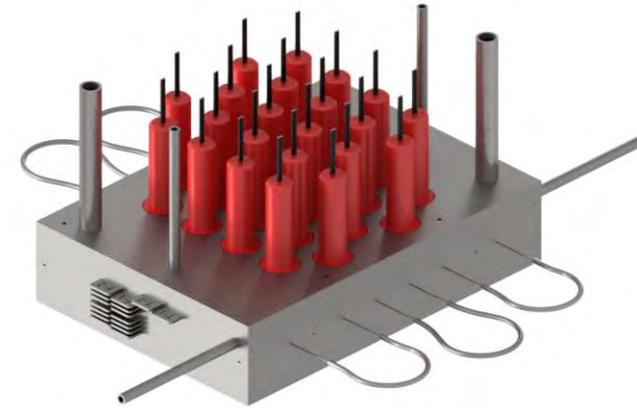
Heat Exchanger	Annular/PCHE Gap	Cycle Efficiency	$\Delta P$ [kPa]	$\Delta T$ [°C]
AFHX (air)	1.9 mm	34.3 %	32.6	51.1
PCHE (air)	1.0 mm	35.3 %	14.2	43.1
PCHE (sCO <sub>2</sub> )	0.6 mm	48.6 %	39.0	15.5



Cycle efficiency vs channel thickness

# sCO<sub>2</sub> Brayton Test Specimen

- Design pressure 20 MPa
  - 16 -1.5 mm 316 SS plates
  - 1- instrumentation layer w/ TC's and FOTS
  - 9.5" x 6.5" x 2"
- 22-130 W cartridge heaters
- Can replicate conditions of the full "wedge"

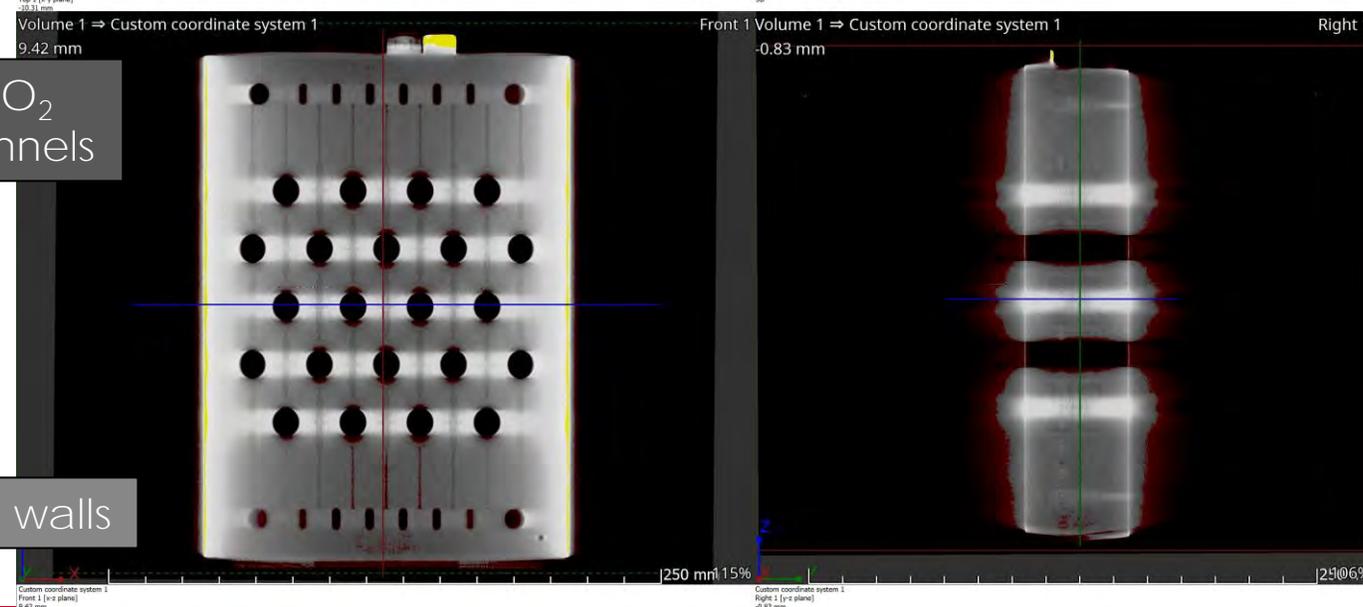
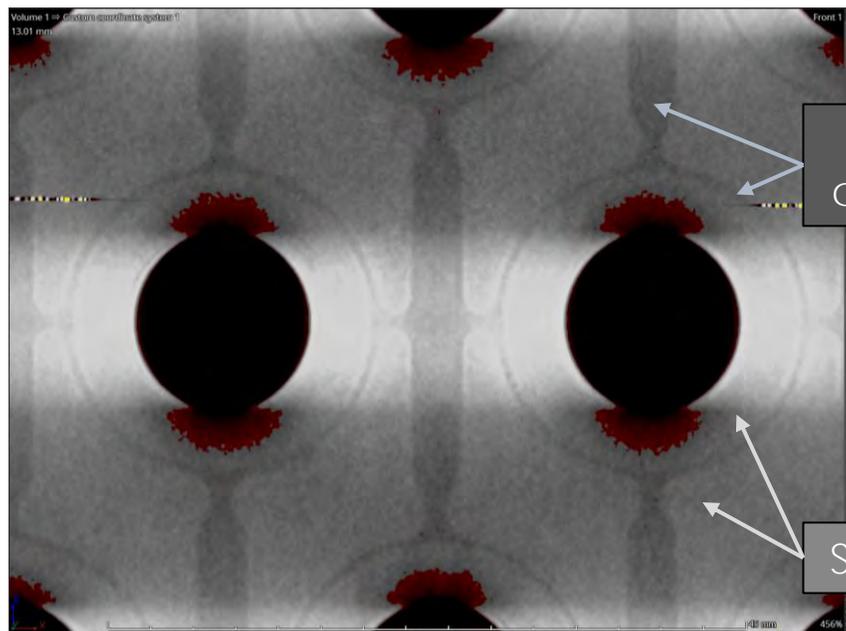
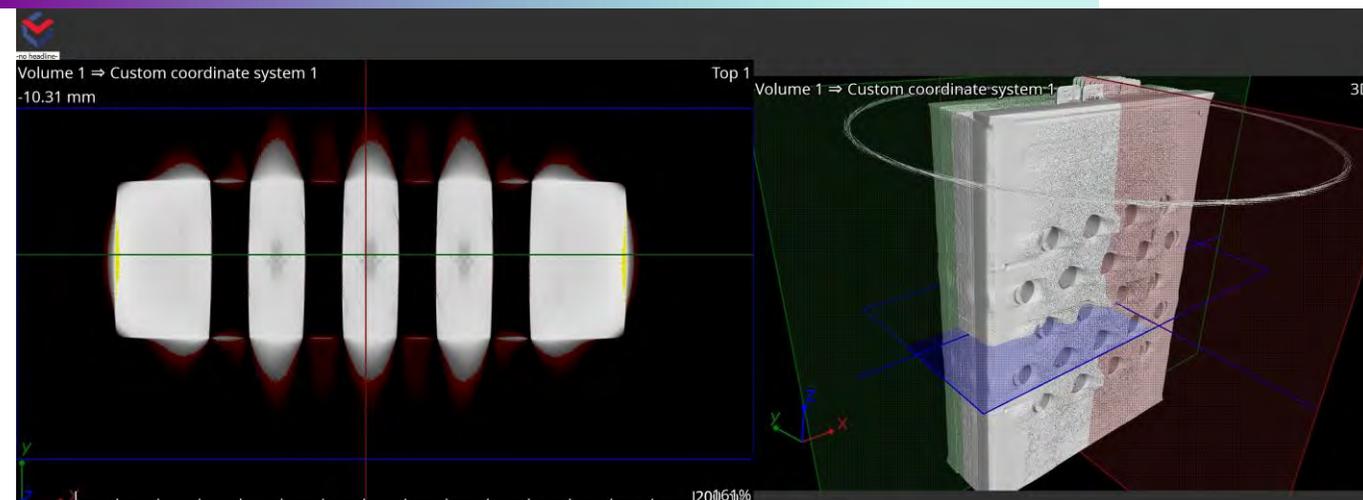


	Test Article	Full Size HX
Power	2.75 kW	5000 kW
Energy density	66 W/in <sup>2</sup>	67 W/in <sup>2</sup>
Cross section	0.08 in <sup>2</sup>	0.08-0.34 in <sup>2</sup>
Mass flow rate	0.06-0.16 kg/s	24.5 kg/s



# sCO<sub>2</sub> Brayton Test Specimen

- CT Scan
- sCO<sub>2</sub> test specimen
- Can visualize the flow channels
- Get dimensions before machining



# Next Steps – HPIHX



- Conclusions
  - Experimental data was used to validate HHXT model for PCHE
  - With the validated model cycle efficiency for the PCHE showed improvement over AFHX
  - Increased heat pipe length increase efficiency ~ 3-5%
  - PCHE channel geometry optimized for sCO<sub>2</sub> Brayton cycle with CFD
  - Increased cycle efficiency over air Brayton cycle by ~ 13%
- Next steps
  - Manufacture and test sCO<sub>2</sub> test article
  - Demonstrate performance and model validity
  - Test with fiber optic temperature sensors
  - Final report





# Thermal Hydraulics Laboratory

## Questions?